

IN THE SPECIFICATION

Please amend page 1, line 1, as follows:

TITLE OF THE INVENTION A turbine vane cooled by a reduced cooling air leak

Please amend the paragraph beginning at page 4, line 27, with the following rewritten paragraph:

Brief description of the drawings

Other characteristics and advantages of the present invention appear from the following description made with reference to the accompanying drawings which show an embodiment having no limiting character, and in which:

- Figure 1 is a perspective view of a turbine nozzle vane of the invention;
- Figure 2 is a detail view of Figure 1 showing a leakage zone in a first embodiment of the invention;
- Figure 3 is a detail view of Figure 1 showing a leakage zone in a second embodiment of the invention;
- Figure 4 is a magnified view of a portion of Figure 3;
- Figure 5 is a perspective view of a prior art turbine nozzle vane; and
- Figure 6 is a section view of the Figure 5 vane; and
- Figure 7 is a detail view of Figure 1 showing a leakage zone in another embodiment of the invention.

Please amend the paragraph beginning at page 6, line 1, with the following rewritten paragraph:

In the invention, it is proposed to establish head losses in the leakage zone so as to reduce the flow rate of cooling air passing therethrough. For this purpose, and as shown in the embodiments of Figures 2 and 3, the sliding zone 16A at the bottom end of the liner 18 in register with the inner platform 16 is provided with a recess 30 formed in all or part of the periphery of said zone in the form of a small notch or groove of shape and dimensions that depend on the desired leakage rate. This recess is preferably circularly symmetrical (i.e. circumferential), e.g. being annular or elliptical, as shown in FIGS. 2 and 7, respectively. As also illustrated in the figures, in one of the embodiments of the present invention, the circular groove is disposed along an azimuthal direction of the end of the liner.